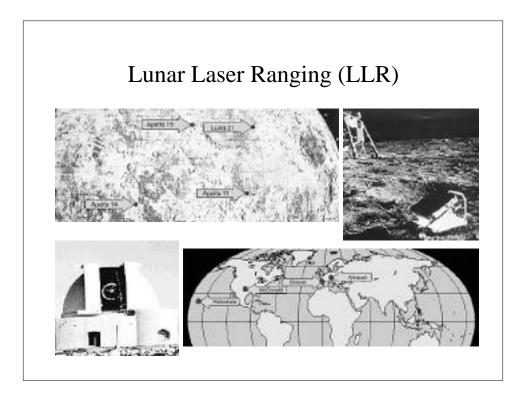
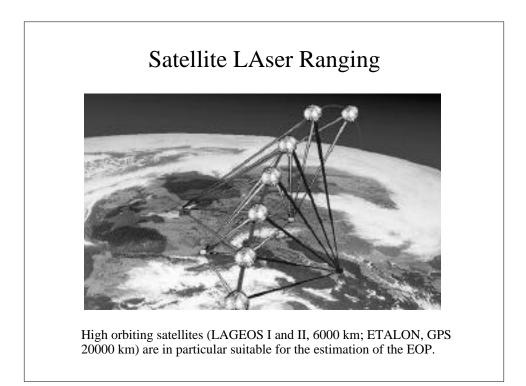


Employing state-of-the-artgeophysical models and a sophisticated least squares parameter estimation process the delays are adjusted and the relevant parameters like *station coordinates, baseline lengths, radio source positions* and Earth rotation parameters (*precession, nutation, polar motion and Universal Time*) are determined.



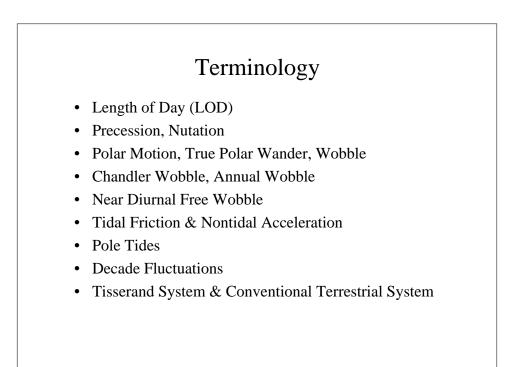
# Lunar Laser Ranging (LLR)

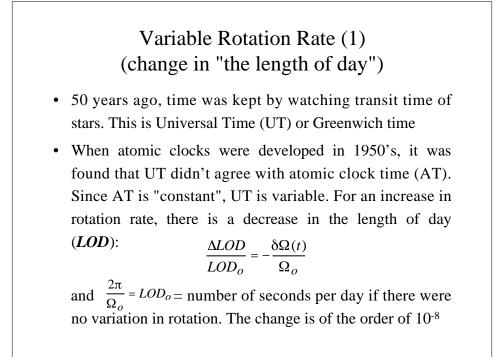
- The Earth-Moon distance varies between 56 and 64 Earth radii.
- The observatories fix a reference frame on Earth & the reflector arrays on the Moon.
- By analyzing the observations (round trip travel times of laser pulses), one can determine the Earth Rotation, parameters describing the dynamics of the Earth-Moon system and relativistic quantities.

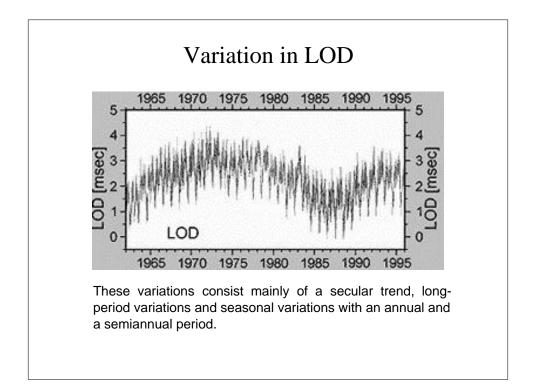


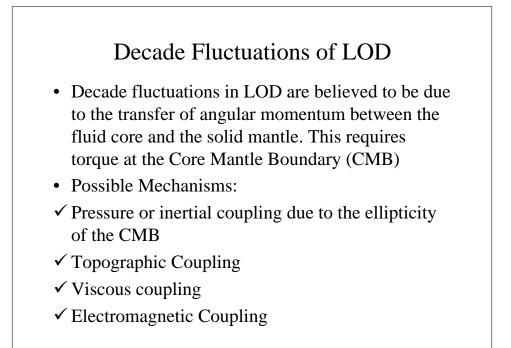
# Satellite LAser Ranging (SLAR)

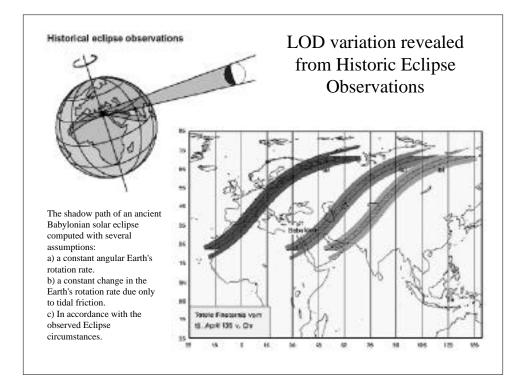
- The satellite's state at any time is the set of its position and velocity as well as parameters which appear in the dynamic acceleration model or the measurement model. The motion of the satellite is governed by a differential equation system that is integrated to determine the state at any later time. Errors in initial values and models necessitate the introduction of observations to the real satellite's motion to obtain a better trajectory.
- Observation equations taking the partial derivatives and the difference of calculated and observed distance from the orbit determination can be solved for the parameters of interest.
- For EOP, a global network of SLR stations renders the estimation of the point about which the stations are rotating during the observation time. A transformation to an earth-fixed reference frame yields the pole coordinates.

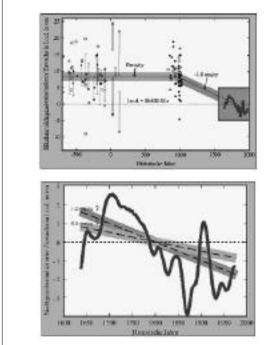










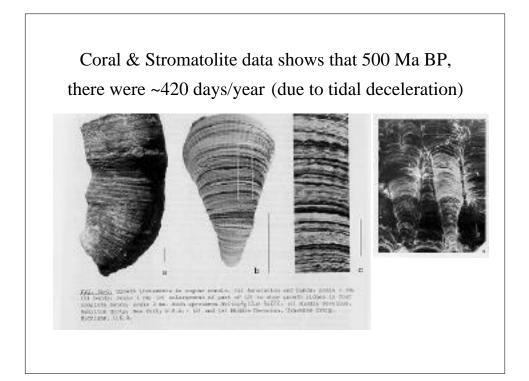


### LOD Variations

Non-tidal changes of the Earth's rotation rate as observed from telescopic data. Accurate astrometrical satellite data exists only for the last 25 years. Secular changes of the Earth's rate of rotation can only be determined with the help of medieval and ancient astronomical observations.

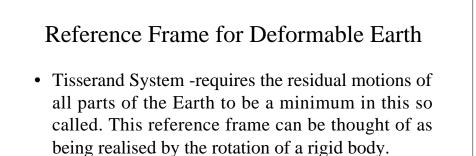
According to Stephenson & Morrison a long-term fluctuation in the l.o.d. with a semi-amplitude of some 4 ms and a period of 1500 yr may exist.

A careful critical review of medieval Arab eclipse records shows that the historicalsources are not in contradiction with a constant secular change in the Earth's rotation rate.

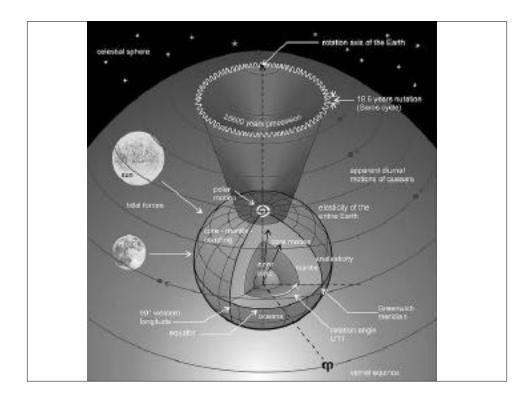


### Variable Rotation /Spin Rate (2)

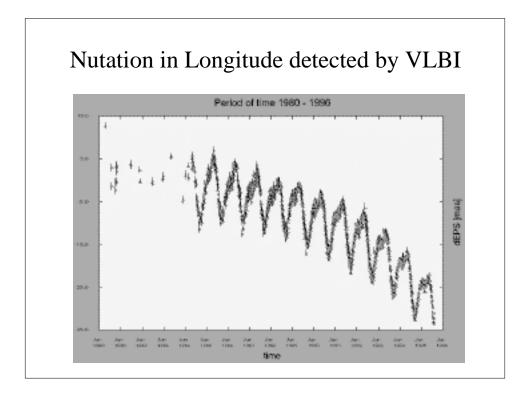
- Paleo-rotation data: there were ~420 days/year during Cambrian period ~500 Ma ago(Tidal Friction)
- Linear increase of ~2 ms per century (Tidal Friction & Postglacial Rebound)
- Decade fluctuations of about 4-5 ms over 20-30 years
- Annual and shorter period fluctuation (atmosphere)
- Modern techniques: VLBI (very-long-baseline Interferometry), LLR (lunar laser ranging) & SLR (satellite laser ranging). Accuracy: better than 0.1 ms for averages over 3-5 days. (Results improve when longer averaging times are used.)

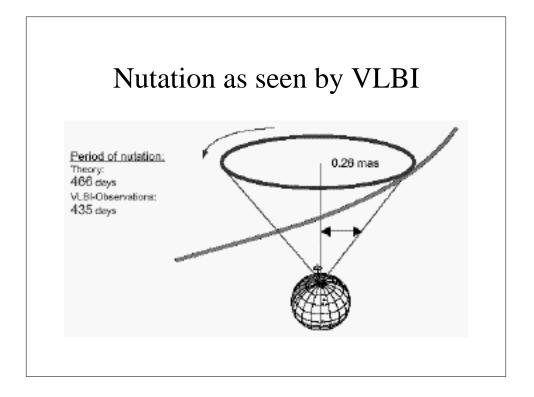


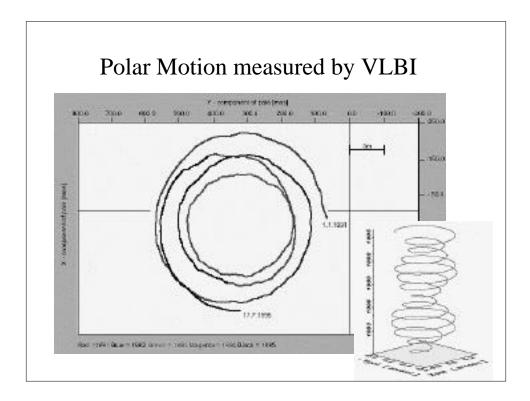
• Conventional Terrestrial System (CTS) is a rigid reference frame co-rotating with the Earth in inertial space with angular velocity  $\Omega$ 

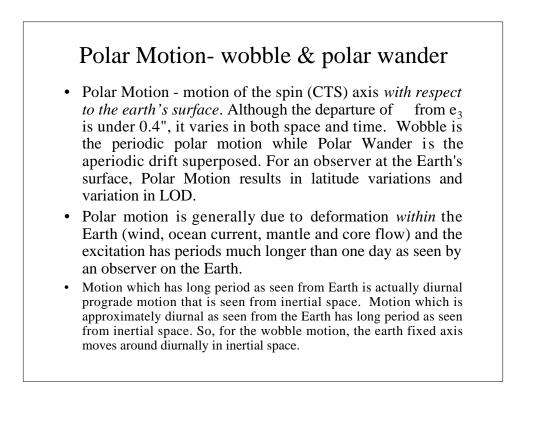


# Nutation & Precession These are motion of the spin (CTS) axis with respect to inertial space (in practice, relative to a quasi-inertial frame tied to the ecliptic and equinox at a certain epoch). Precession generally refers to the slow motion with period of ~26,000 years while Nutation superposes a small nodding motion with a period of 18.6 years and an amplitude of 9.2 seconds of arc. They are caused by the gravitational torques of the Moon and Sun on the spinning Earth's equatorial bulge. (The plane of the Moon's orbit around the Earth is tilted by about 5° from the plane of the Earth's orbit around the Sun.)



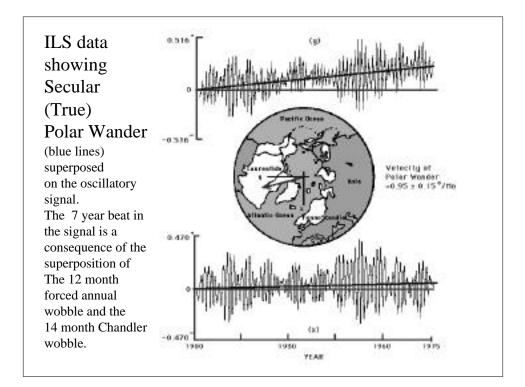


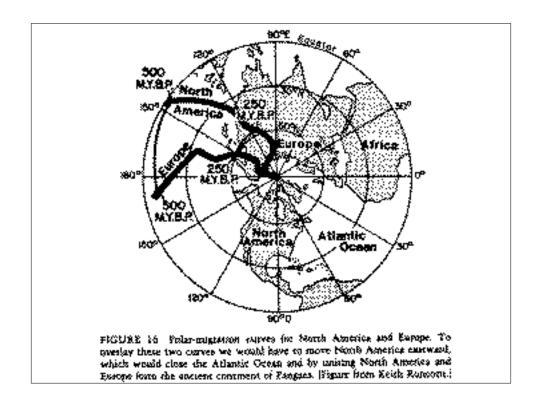


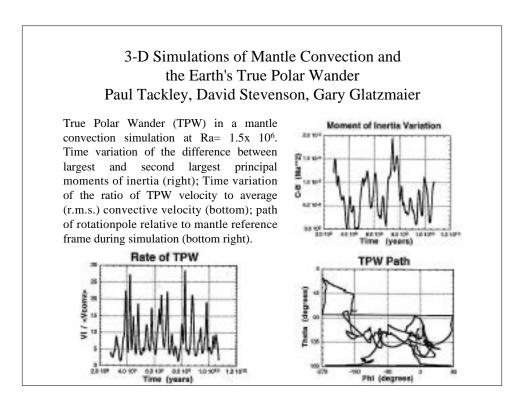


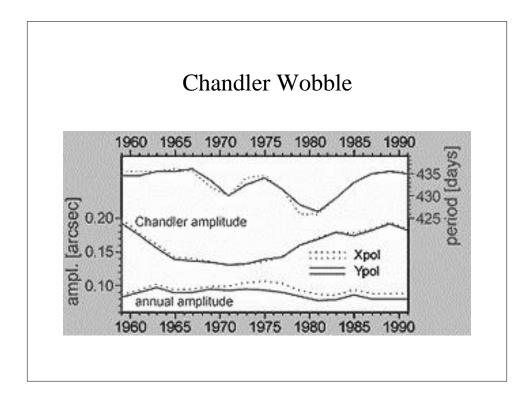
# Polar Motion (continue)

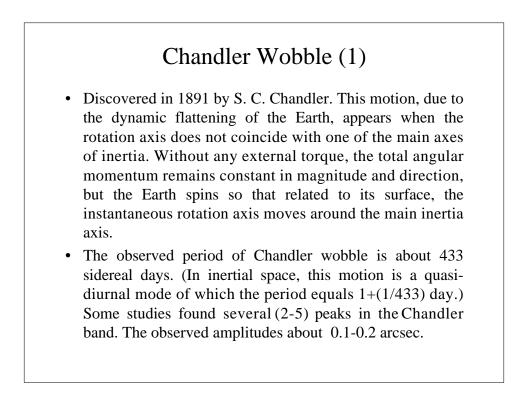
- In fact, nutation cannot occur without some accompanying polar motion or vice versa. For example, a small part of polar motion is the 'dynamical variation of latitude' as traces out a nearly diurnal retrograde circuit in the CTS of amplitude ~0.02", due to the forced nutations.
- Free Chandler wobble (period ~ 433 sidereal days) and forced Annual wobble (amplitude ~0.1 arcsec or 5 m)
- Polar wander rate of ~1°/Ma towards Hudson Bay today (due to postglacial rebound)
- Longer term True Polar Wander & Continental Drift











# Chandler Wobble (2)

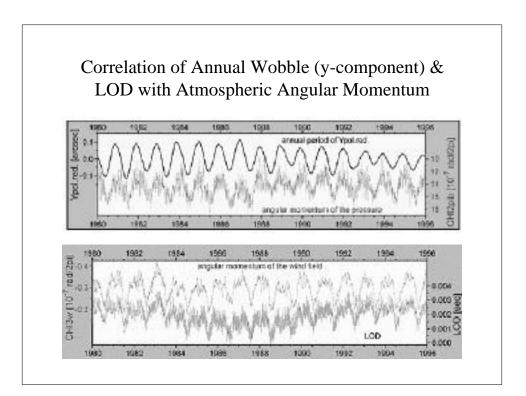
- For a rigid Earth, Euler showed that the pole displacement in the terrestrial frame produces a latitude variation with a period of 305 days.
- If one takes into account of the elasticity of the Earth, then the period increases to 445 days.
- Including the fluid core below the elastic mantle would reduce the period to about 405 days.
- Further inclusion of the small pole tide set up in the oceans by centrifugal force would increase the period by ~30 days
- If one accounts for dissipation in the mantle, core and oceans, then the predicted period would come close to the observed period of about 433 days.

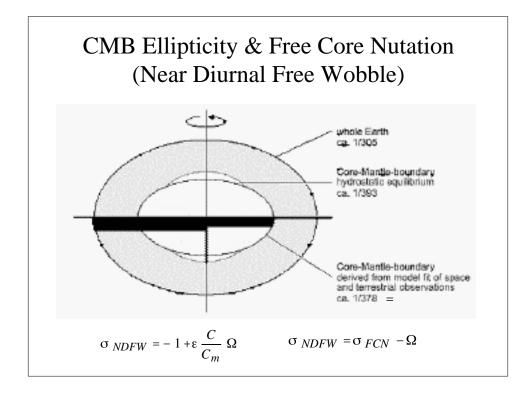
# Chandler Wobble (3)

- Chao (1983) explains the multiple peak structure in the Chandler band with the existence of nonelastic layers in the Earth (e.g. hydrosphere, asthenosphere & outer core) and their coupling with the visco-elastic spheres of the Earth
- The broadening of the Chandler frequency give quality factor Q~179 (i.e. decay time ~68 years), so that the amplitude would quickly dampen to zero unless some mechanism or combination of mechanisms are exciting it
- Plag (1997) hypothesized that the Chandler wobble is a forced, quasi-periodic motion close to a resonance period

# Chandler Wobble (4)

- This free oscillation can be excited by mass redistribution in atmosphere, oceans and mantle (due to earthquakes).
- However, the changes in moment of inertia due to earthquakes are orders of magnitude too low and the occurrence of earthquakes are not frequent enough to sustain Chandler wobble. In the 1990's, attention turned to atmospheric forcing as the main cause of Chandler wobble
- Gross (2000) reports that the principal cause of the Chandler wobble is fluctuating pressure on the bottom of the ocean, caused by temperature and salinity changes and wind-driven changes in the circulation of the oceans. Gross calculated that two-thirds of the Chandler wobble is caused by ocean-bottom pressure changes and the remaining one-third by fluctuations in atmospheric pressure. Apparently, the effect of atmospheric winds and ocean currents on the wobble was minor



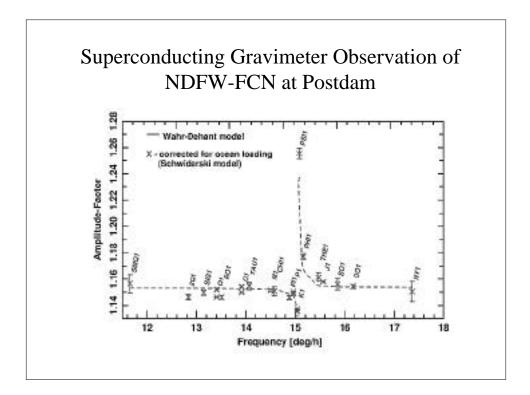


### Near Diurnal Free Wobble (NDFW) & Free Core Nutation (FCN)

- The CMB is elliptical, then the inertia of the core would resist the tilt of the mantle by inertial coupling. Thus, if the rotation axis of the mantle and core become misaligned, then the restoring forces at the elliptical CMB will try to realign the two axis. Because the earth is a fast spinning gyro, the reaction is a damped wobble of the instantaneous rotation axis around the figure axis (NDFW) when observed in the terrestrial reference frame. The moton of the pole is retrograde about the body axis with period ~ 1 sidereal day (differ by about 4 minutes).
- Viewed from the celestial frame (e.g. VLBI) it is called Free Core Nutation (FCN) and has a period of 432 days.

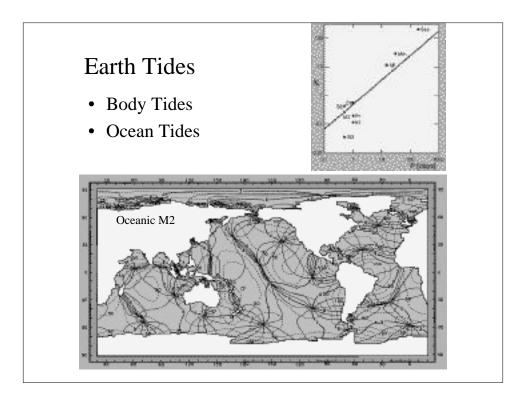
### Near Diurnal Free Wobble (NDFW) & Free Core Nutation (FCN)

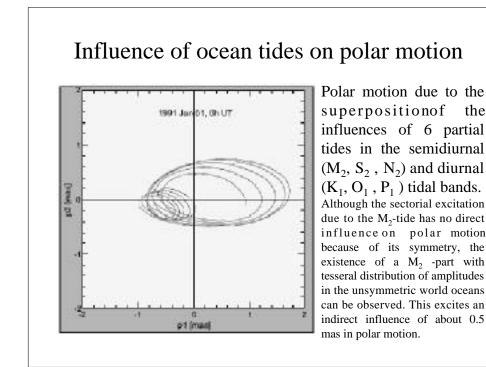
- The closeness of the frequencies of the NDFW mode and the diurnal tidal frequency band means that there is resonance between them. The NDFW resonantly amplifies nearly diurnal tides and annual and semiannual nutations.
- Observing the NDFW / FCN is thus very useful to measure the CMB flattening and to obtain information about the dissipation effect at this interface.
- Existence of the solid inner core leads to 2 additional eigenmodes, an inner core wobble with frequency far outside the diurnal band and a prograde free inner core nutation with its associated wobble.

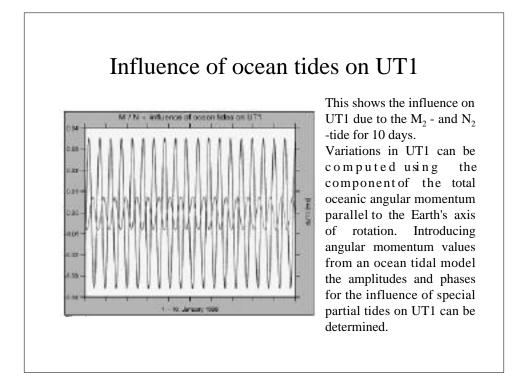


### Near Diurnal Free Wobble (NDFW) & Free Core Nutation (FCN)

- Existence of the solid inner core leads to 2 additional eigenmodes, an inner core wobble with frequency far outside the diurnal band and a prograde free inner core nutation with its associated wobble.
- Unambiguous direct observation of the FCN with an amplitude of 174 µas was achieved by Herring & Dong (1994) in an analysis of 8 years of VLBI data. The wobble amplitude is 400 times smaller and is thus more difficult to measure
- Jiang & Smylie (1995) also claimed the detection of a retrograde nutation with a period of 43120 solar days in the nutation data obtained with VLBI.

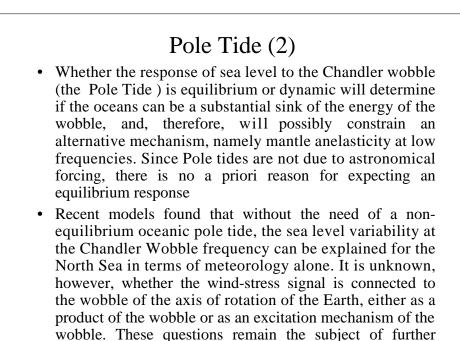






# Pole Tide (1)

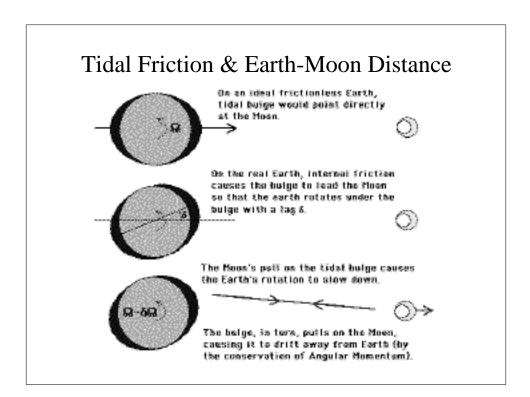
- Tide gauge data shows that in North Sea and Baltic Sea, there is a statistically significant signal at Chandler Wobble frequency, with amplitude of a few cm - i.e. several times larger than expected from the equilibrium theory
- The pole tide is not due to astronomical forcing, as are the luni-solar tides
- Poles Tides are due to changes in the centrifugal forces during Chandler wobble which produce a signal in sea level at the same frequency



research

# Tidal Friction (1)

- The Moon deforms the Earth and oceans into the ellipsoidal shape
- The orientation of the ellipsoidal bulge is fixed with respect to the Moon, while the Earth rotates at 1 cycle/day relative to the bulge. The resulting lunar tides are time dependent, with frequencies equal to integral multiples of 1 cycle/day, modulated by the frequencies of the lunar orbit (e.g. 1 cycle per 27.7 days and 1 cycle per 13.7 days)



# Tidal Friction (2)

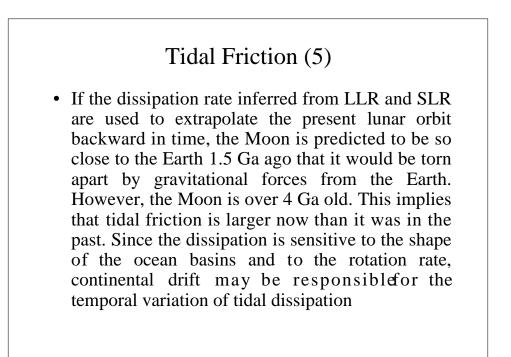
- If there were no energy dissipation in the Earth and oceans, the ellipsoidal bulge should point towards the Moon. However, there is some dissipation and the maximum tidal uplift occurs shortly after the Moon is overhead, and the bulge leads the Earth-Moon vector by a small angle 3. (From , seismic attenuation or quality factor Q can been estimated to be ~20 and is lower than most seismic estimates.)
- The Moon's gravitational force acts on the tide bulge to produce a clockwise torque on the Earth, opposite to its rotation. Thus, there is an increase in LOD (Tidal Deceleration) of ~ 2.3 ms/century
- There is a similar, although smaller, effect from the sun

### Tidal Friction (3)

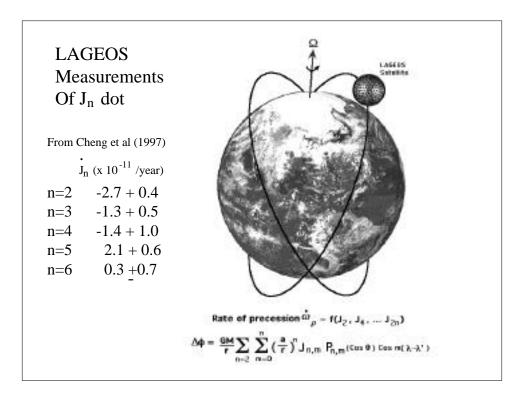
- Most of the tidal energy dissipation is believed to occur in the oceans. It is still not entirely clear whether most of the dissipation occurs in shallow seas or in deep oceans
- The Earth's tidal bulge causes a counter-clockwise torque in the direction of the Moon's motion, thus increasing its angular momentum. The increase in lunar angular momentum causes the Moon to move farther away from the Earth at a rate of ~4 cm/a and to increase its orbital period. This increase in period has been determined accurately from LLR data.

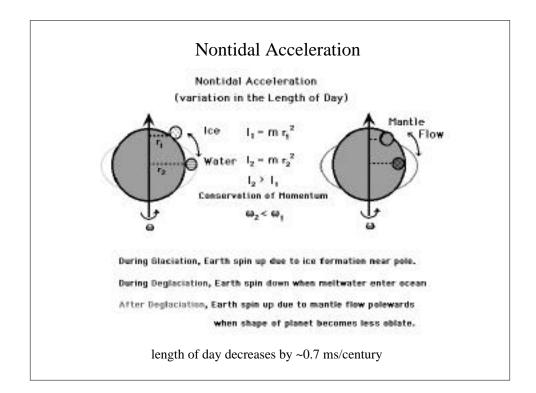
# Tidal Friction (4)

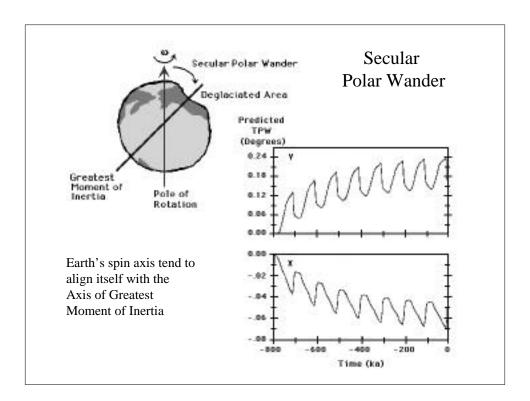
- The tidal bulge perturbs the orbit of a satellite, thus by ranging to satellites such as LAGEOS, the lag angle can be determined. The lunar torque on the Earth can also be determined from and thus the change in LOD can be calculated. This predicted increase in the LOD is about 25% larger than that implied by the historical eclipse record (~1.7 ms/century)
- The reason for this discrepancy is due to Postglacial Rebound, which cause a net transfer of mantle material toward the poles. This redistribution of internal mass decreases the earth's polar moment of inertia and give rise to the observednon-tidal acceleration of the Earth. Postglacial rebound also causes a net polar wander towards Hudson's Bay today

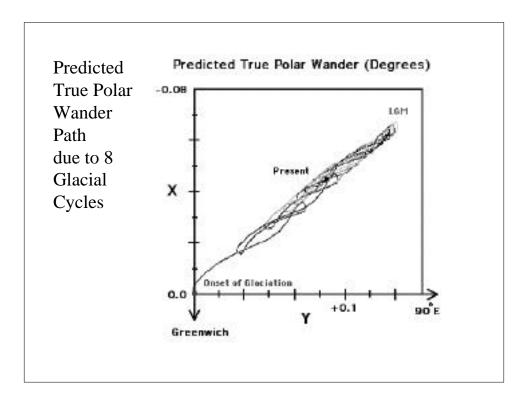


Source	$(10^{-10}/\text{year})$
Currot (1966)	0.7 + 0.3
Muller & Stephenson (1975)	1.5 + 0.3
Morrison (1973)	2.9 + 0.6
Lambeck (1977)	0.69 + 0.3









Euler Equation for the conservation of angular momentum:  

$$\frac{d}{dt} \left( J_{ij} \omega_j \right) + \varepsilon_{ijk} \omega_j J_{kl} \omega_l = 0$$
In the unperturbed state,  $\begin{bmatrix} J_{ij} \end{bmatrix} = \begin{pmatrix} A & 0 & 0 \\ 0 & B & 0 \\ 0 & 0 & C \end{pmatrix}$  where  $B = A$   
with angular velocity vector :  $\begin{pmatrix} 0 \\ 0 \\ \Omega \end{pmatrix}$   
In the perturbed state, the angular velocity vector is  $\begin{pmatrix} m_1 \\ m_2 \\ \Omega + m_3 \end{pmatrix}$   
and  $\begin{bmatrix} J_{ij} \end{bmatrix} = \begin{pmatrix} A + I_{11} & I_{12} & I_{13} \\ I_{12} & A + I_{22} & I_{23} \\ I_{13} & I_{23} & C + I_{33} \end{pmatrix}$ 

**Liouville's Equation**- linearized Euler Equations  $\frac{i}{\sigma_r} \dot{m} + m = \Psi \quad \text{and} \quad \dot{m}_3 = \dot{\psi}_3$ where  $m = m_1 + im_2$  and  $\sigma_r = \Omega \frac{(C-A)}{A}$  is the Chandler frequency of a rigid earth, and the Excitation Functions are (Munk & MacDonald 1960):  $\Psi = \psi_1 + i\psi_2$   $\psi_1 = \frac{I_{13}}{(C-A)} + \frac{\dot{I}_{23}}{\Omega(C-A)}$  $\psi_2 = \frac{I_{23}}{(C-A)} - \frac{\dot{I}_{13}}{\Omega(C-A)}$   $\psi_3 = \frac{-I_{33}}{C}$ 

$$\Psi = \Psi \stackrel{Rot}{} + \Psi \stackrel{Load}{}$$
First consider the Excitation due to Rotation:  
following Lambeck(1980), the centrifugal potential is:  
$$\chi = \frac{1}{3}\omega^2 r^2 + \frac{GM}{r} \frac{a}{r} \frac{2}{r} \frac{2}{m=0} [C_{2m} \cos(m\phi) + S_{2m} \sin(m\phi)] P_2^m (\cos\theta)$$
where  
$$C_{20} = \frac{a^3}{6GM} (\omega_1^2 + \omega_2^2 - 2\omega_3^2) - k_2^T$$

$$C_{21} = -\frac{a^3}{3GM} \omega_1 \omega_3 - k_2^T \qquad C_{22} = \frac{a^3}{12GM} (\omega_2^2 - \omega_1^2) - k_2^T$$

$$S_{21} = -\frac{a^3}{3GM} \omega_2 \omega_3 - k_2^T \qquad S_{22} = -\frac{a^3}{6GM} \omega_1 \omega_2 - k_2^T$$

Using MacCullagh's formula (Munk & MacDonald 1960):  

$$I_{13}^{Rot} = \frac{k_2^T}{k_f} \quad m_1 (C - A) \qquad I_{23}^{Rot} = \frac{k_2^T}{k_f} \quad m_2 (C - A)$$
where  

$$k_f = \frac{3G}{a^5 \Omega^2} (C - A)$$
Thus  

$$\psi_1^{Rot} = \frac{k_2^T}{k_f} \quad m_1 + \frac{m_2}{\Omega} = \frac{k_2^T}{k_f} \quad m_1$$

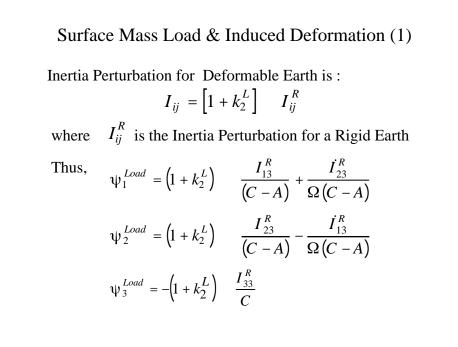
$$\psi_2^{Rot} = \frac{k_2^T}{k_f} \quad m_2 - \frac{m_1}{\Omega} = \frac{k_2^T}{k_f} \quad m_2$$
and  

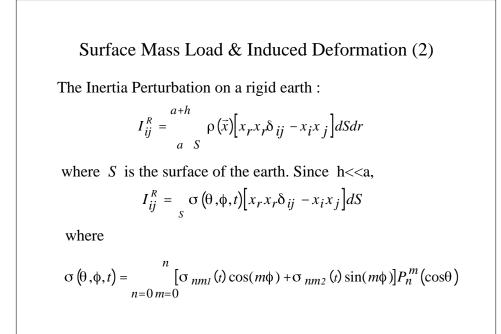
$$\Psi^{Rot} = \psi_1^{Rot} + i\psi_2^{Rot} = \frac{k_2^T}{k_f} \quad m_1 + im_2$$

Thus, Liouville's Equation becomes  $\frac{i}{\sigma_r} \dot{m} + m - \frac{k_2^T}{k_f} \quad m = \Psi^{Load}$  Polar Motion  $\dot{m}_3 = \dot{\psi}_3$  Length of Day Variation As shown in Wu & Peltier (1984), the first term in the equation for Polar Motion contains the contribution of the Chandler Wobble. Thus

$$\left[1-k_2^T/k_f\right] \quad \boldsymbol{m} = \Psi^{Load}$$

where **m** only contains secular variations.





### Surface Mass Load & Induced Deformation (3)

Using the orthogonal relationship for un-normalized Associated Legendre function :

$$P_n^m(\cos\theta)\frac{\cos(m\phi)^2}{\sin(m\phi)}dS = \frac{4\pi(n+m)!}{(2n+1)(n-m)!(2-\delta_{0m})}$$

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Therefore:

$$I_{13}^{R} + iI_{23}^{R} = -\frac{4}{5}\pi a^{4} \left[\sigma_{211} + i\sigma_{233}^{R}\right]$$
$$I_{33}^{R} = \frac{8}{3}\pi a^{4} \sigma_{001} - \frac{\sigma_{201}}{5}$$

since mass is conserved in GIA  $\sigma_{001} = 0$  therefore

$$I_{33}^{R} = -\frac{8}{15}\pi a^{4}\sigma_{201}$$

Summary of Equations in Time Domain  $\begin{bmatrix} 1 - k_2^T / k_f \end{bmatrix} \quad \{m_1 + i \ m_2\} = \psi_1^{Load} + i\psi_2^{Load}$   $\psi_1^{Load} = -\frac{4}{5}\pi a^4 \frac{\left(1 + k_2^L\right)}{\left(C - A\right)} \quad [\sigma_{211}(t) + \dot{\sigma}_{212}(t) / \Omega]$   $\psi_2^{Load} = -\frac{4}{5}\pi a^4 \frac{\left(1 + k_2^L\right)}{\left(C - A\right)} \quad [\sigma_{212}(t) - \dot{\sigma}_{211}(t) / \Omega]$   $\dot{m}_3 = \dot{\psi}_3^{Load}$   $\psi_3^{Load} = \frac{8}{15} \frac{\pi a^4}{C} \left(1 + k_2^L\right) \quad \sigma_{201}(t)$ 

Variation in LOD and 
$$J_2$$
-dot $\dot{m}_3(t) = \frac{8}{15} \frac{\pi a^4}{C} \frac{\partial}{\partial t} + k_2^L - \sigma_{201}(t)$ Expanding the potential perturbation as surface mass density: $\phi(\theta, \psi, t) = ag \prod_{n=0}^{n} [J_{nm1}(t)\cos(m\psi) + J_{nm2}(t)\sin(m\psi)]P_n^m(\cos\theta)$  $= \frac{4\pi a^3 g}{M} \prod_{n=0}^{n} \frac{1+k_n^L}{2n+1} [\sigma_{nm1}(t)\cos(m\psi) + \sigma_{nm2}(t)\sin(m\psi)]P_n^m(\cos\theta)$ Comparing the two : $J_{nmi}(t) = \frac{4\pi a^2}{M} \frac{[1+k_n^L(t)]}{(2n+1)} - \sigma_{nmi}(t)$ Therefore:

Love Number Approach for Polar Wander (1): Load and  $k_2^L(s) = k_2^{LE} + \sum_{i=1}^{N} \frac{r_i}{s - s_i} = -1 + l_s + s \sum_{i=1}^{N} \frac{r_i/s_i}{s - s_i}$   $k_2^T(s) = k_2^{TE} + \sum_{i=1}^{N} \frac{t_i}{s - s_i} = k_2^T(0) + s \sum_{i=1}^{N} \frac{t_i/s_i}{s - s_i}$ defining  $g_j = \frac{t_j/s_j}{N(t_i/s_i)}$   $k_f = \frac{\sigma_f}{\sigma_o} \sum_{i=1}^{N} (t_i/s_i)$   $\frac{\sigma_o}{\sigma_f} = \frac{k_2^T(0) - k_2^{TE}}{k_2^T(0)}$   $Q_{N-1}(s) = \sum_{i=1}^{N} \frac{g_j}{s - s_i} \sum_{j=1}^{N} (s - s_j) = \sum_{j=1}^{N-1} (s + \lambda_j)$   $q(s) = s \sum_{j=1}^{N-1} (s + \lambda_j) - \sum_{i=1}^{N} (s - s_i)$  $R_j(s) = \sum_{j=1}^{N-1} (s + \lambda_j) - \sum_{i=1}^{N} (s - s_i)$ 

Love Number Approach for Polar Wander (2):  

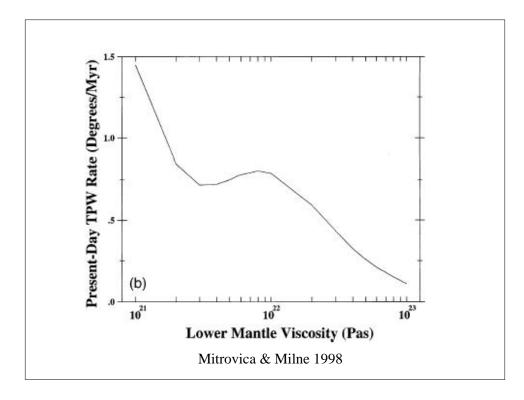
$$\dot{m}_{j}(t) = \frac{\Omega}{A\sigma_{o}} D_{1}\dot{I}_{j3}^{R}(t) + D_{2}I_{j3}^{R}(t) + \sum_{i=1}^{N-1} E_{i} \frac{\partial}{\partial t} \left\{ e^{-\lambda_{i}t} I_{j3}^{R}(t) \right\}$$

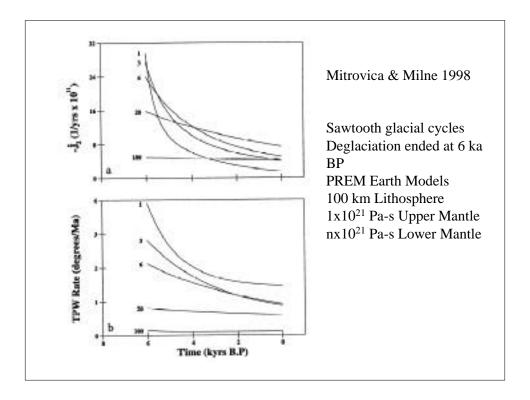
$$\left\langle m_{j}(t) \right\rangle = \frac{\Omega}{A\sigma_{o}} D_{1}I_{j3}^{R}(t) + D_{2} \int_{t'=0}^{t} I_{j3}^{R}(t')dt' + \sum_{i=1}^{N-1} E_{i} \left\{ e^{-\lambda_{i}t} I_{j3}^{R}(t) \right\}$$

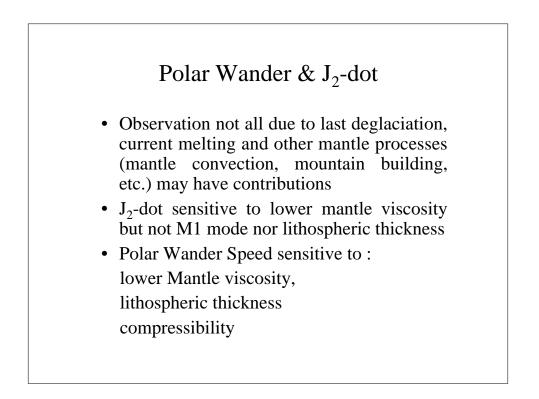
$$D_{1} = l_{s} - \sum_{i=1}^{N} (r_{i}/s_{i}) = 1 + k_{2}^{LE}$$

$$D_{2} = -l_{s}q(0) / \sum_{i=1}^{N-1} \lambda_{i}$$

$$E_{i} = \frac{l_{s}q(-\lambda_{i})}{\lambda_{i}} + \sum_{j=1}^{N} \frac{r_{j}R_{j}(-\lambda_{i})}{s_{j}} / \sum_{k=1}^{N-1} (\lambda_{k} - \lambda_{i})$$







### Some References

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#### <u>Wu (1982)</u>

Prove that the inertia perturbation due to deformation:  $I_{ij}^D = k_2^L - I_{ij}^R$ By definition:  $I_{ij}^{D} = \int_{V} \rho_1 \left( x_r x_r \delta_{ij} - x_i x_j \right) dV + \int_{V} \rho_0(a) \left( a^2 \delta_{ij} - x_i x_j \right) u(a) dS$ 

Below, we will only consider  $I_{33}^D$  - the other components will follow similarly. For  $I_{33}^D$ , the above becomes:  $I_{33}^D = \rho_1 \left( x_1^2 + x_2^2 \right) dV + \rho_0 \left( x_1^2 + x_2^2 \right) u(a) dS$ 

The gravitational potential can be decomposed into the direct contribution from the load and internal mass redistribution:  $\phi_1 = \phi_2 + \phi_3 = (1 + k_n) \Phi_{2,n} P_n^m (\cos \theta)$ . So that  ${}^2\phi_2 = 0$  and

$$^{2}\phi_{3} = 4\pi G\rho_{1}.$$

Now from Green's identity:

 $\int_{V} \left( Z ^{2}Y - Y ^{2}Z \right) dV = \int_{S} \left( Z\hat{n} ^{-}Y - Y\hat{n} ^{-}Z \right) dS$ using Y =  $\phi_3$  and Z =  $x_1^2 + x_2^2 = \frac{2}{3}r^2 \left[ P_0^0(\cos \theta) - P_2^0(\cos \theta) \right]$ where  ${}^{2}Z = \frac{1}{r^{2}}\frac{\partial}{\partial r} r^{2}\frac{\partial Z}{\partial r} + \frac{1}{r^{2}\sin\theta}\frac{\partial}{\partial\theta}\sin\theta\frac{\partial Z}{\partial\theta} = 4$ 

$$\hat{\mathbf{n}} \quad Z = \frac{\partial}{\partial r} Z = \frac{4}{3} r \left[ 1 - P_2^0(\cos \theta) \right] = \frac{2}{r} Z$$

we get:  $I_{33}^D = \frac{1}{4\pi G} 4 \int_{V} \phi_3 dV + \frac{2}{3}a^2 \int_{S} (P_0^0 - P_2^0)\Gamma(a) dS$ 

where  $\Gamma = \frac{\partial}{\partial n} \phi_3 - \frac{2}{3} \phi_3 + 4\pi G \rho_0 u$ 

In the transformed s-domain, expanding:  $\phi_3 = \Phi_{3,n}(r)P_n^0(\cos\theta), \ u = U_n(r)P_n^0(\cos\theta),$  $\Gamma = \prod_{n} \Gamma_n(r) P_n^0(\cos\theta)$ then  $\int_{V} \phi_{3} dV = 4\pi \int_{0}^{a} r^{2} \Phi_{3,n}(r) dr$ 

and 
$$\int_{S} (P_0^0 - P_2^0) \Gamma(a) dS = 4\pi a^2 [\Gamma_0(a) - \frac{1}{5}\Gamma_2(a)]$$

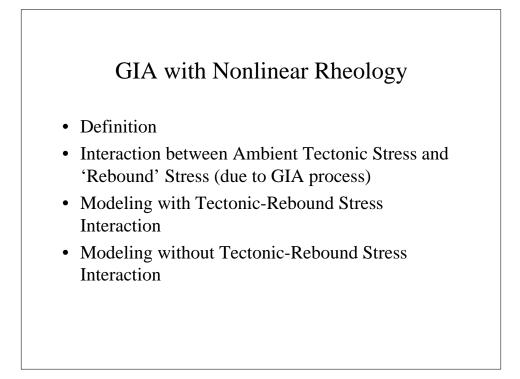
thus, 
$$I_{33}^D = \frac{4}{G} \frac{a}{0} r^2 \Phi_{3,0}(r) dr + \frac{a^2 M}{g} \frac{2}{3} \Gamma_0(a) - \frac{2}{15} \Gamma_2(a)$$

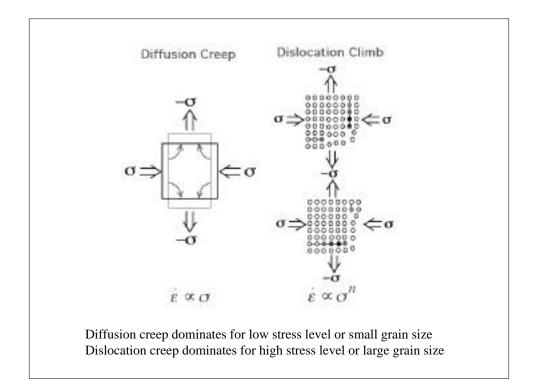
In order to express the  $\Gamma_n$  in terms of  $k_n^L$  use the definition of the gradient of potential:

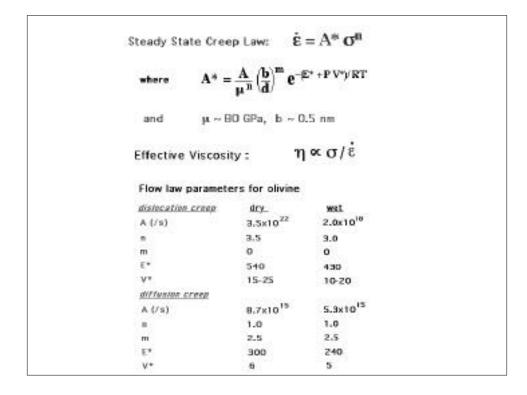
$$Q_n = \frac{\partial}{\partial r} \Phi_{3,n} + \frac{(n+1)}{a} \Phi_{3,n} + 4\pi G \rho_0 U_n$$

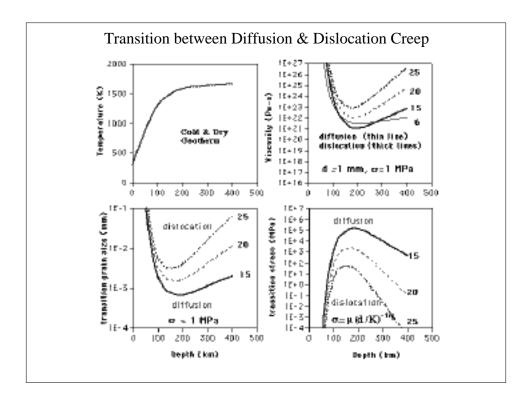
and by comparing with the definition of  $\Gamma_n$ , we see that  $\Gamma_n(a) = Q_n(a) - \frac{(n+3)}{a} \Phi_{3,n}(a)$ Since  $Q_n(a) = 0$ ,  $\Gamma_n(a) = -\frac{(n+3)}{a} \Phi_{3,n}(a) = (n+3) \frac{g}{M} k_n^L L_n$ where  $L_n = a^2 \quad \sigma(\theta, \varphi) P_n^0(\cos\theta) dS = \frac{4\pi a^2}{(2n+1)} \sigma_{n\,01}$  and  $I_{33}^R = -\frac{2}{3} a^2 L_2$ Thus,  $I_{33}^D = \frac{4}{G} \frac{a}{0} r^2 \Phi_{3,0}(r) dr + 2a^2 k_0^L L_0 - \frac{2}{3} a^2 k_2^L L_2$ 

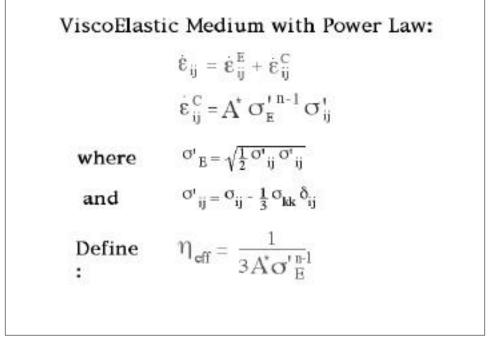
For glacial loading events, the n=0 response is never excited because  $\sigma_{001} = L_0 = 0$ , thus in the transformed s-domain:  $I_{33}^D = -\frac{2}{3} a^2 k_2^L L_2 = k_2^L I_{33}^R$ Other components can be derived similarly.

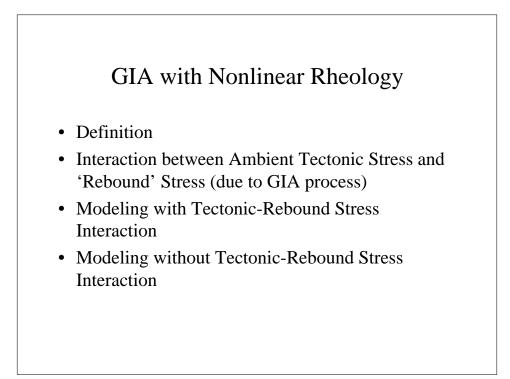




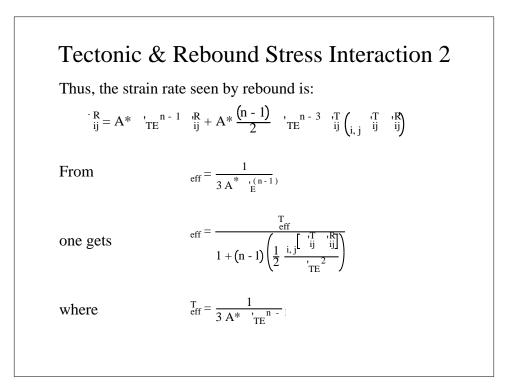








## Tectonic & Rebound Stress Interaction 1 If both rebound & tectonic stresses are present $'_{ij} = '_{ij}^{T} + '_{ij}^{R}$ Substituting this into the creep law $'_{ij}^{C} = A^{*} '_{E}^{n-1} '_{ij}$ where $'_{E} = \sqrt{\frac{1}{2} '_{ij} '_{ij}}$ gives $'_{E} = '_{TE} + \frac{1}{2} \frac{i, j ('_{ij}^{T} '_{ij}^{R})}{'_{TE}}$



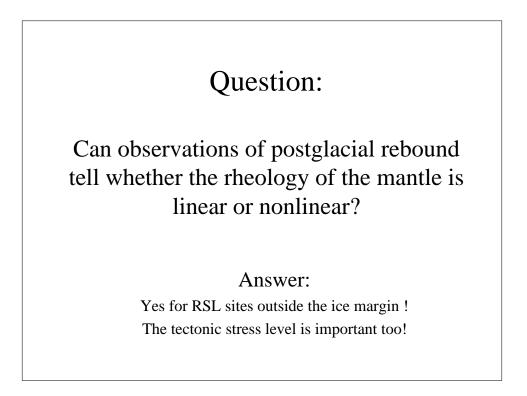
#### Tectonic & Rebound Stress Interaction 3

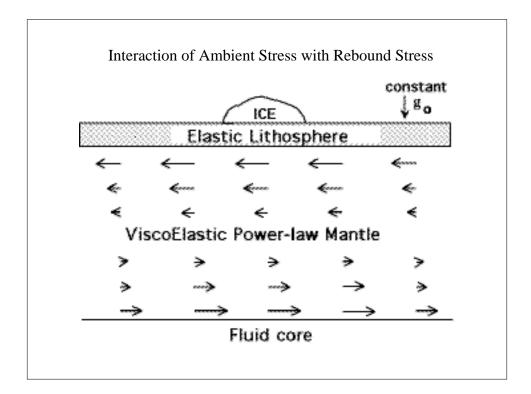
Consider the case when rebound & tectonic stress are orthogonal  $i_{i,j} i_{j}^{T} i_{j}^{R} = i_{kl}^{T} i_{kl}^{R}$ 

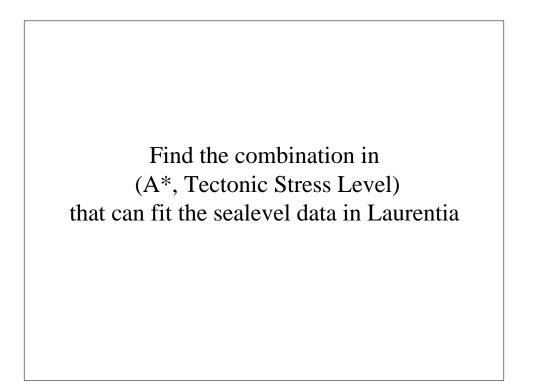
then

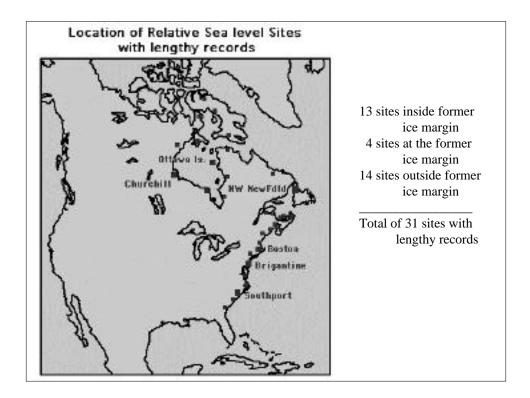
Thus, although the creep law is nonlinear, rebound only 'sees' a linear creep law with the effective viscosity dependent on the tectonic stress distribution.

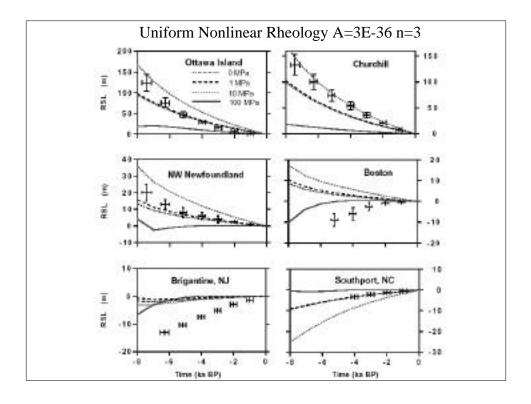
Also, the creep law seen by the ij-th component is different from the kl-component - thus rebound sees a linear but anisotropic creep law.

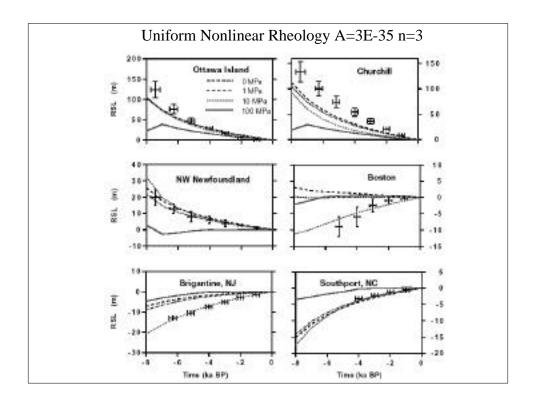


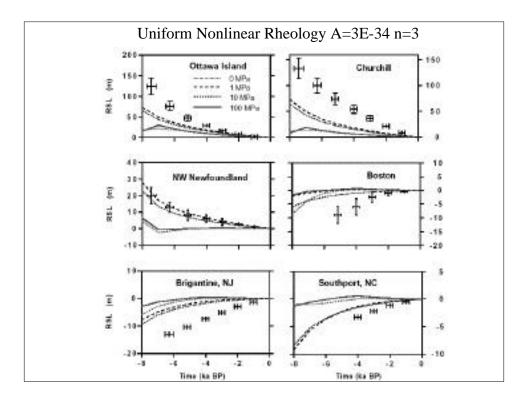








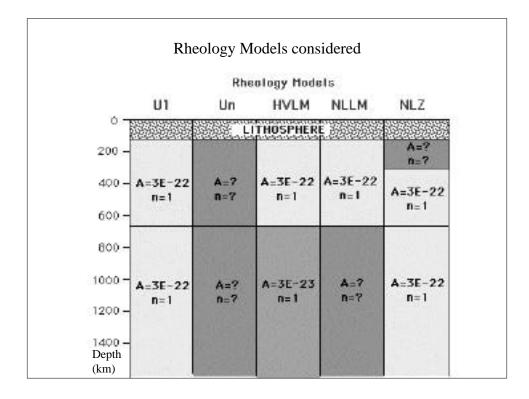


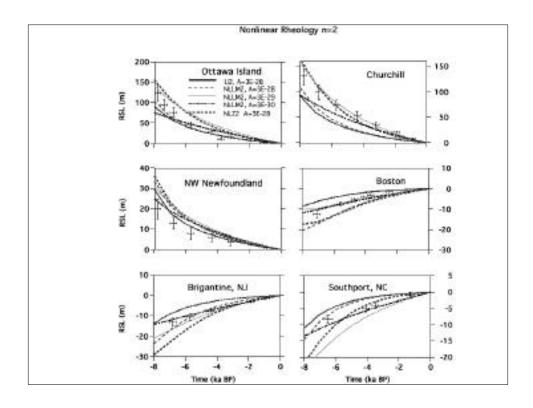


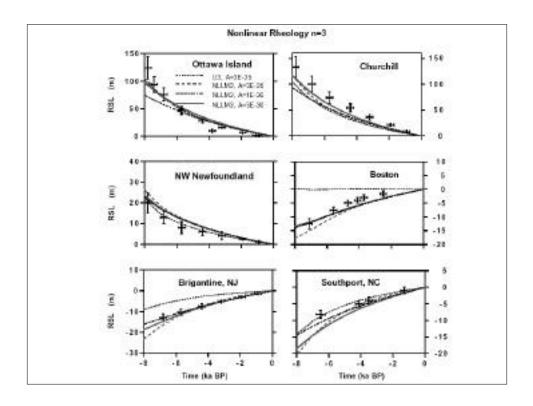
# Summary for the case with Tectonic-Rebound Stress Interaction:

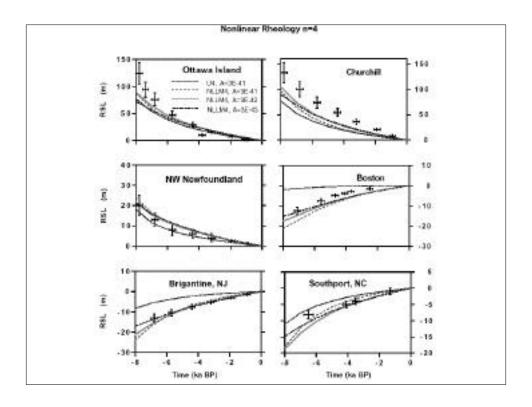
- RSL data just outside the ice margin can distinguish linear mantle from nonlinear mantle
- RSL data inside the ice margin can be explained by both linear and nonlinear rheology
- Nonlinear Uniform Mantles with A=3x10<sup>-35</sup>, n=3 and Tectonic Stress Level ~ 10 MPa can explain the RSL data in and around Laurentia

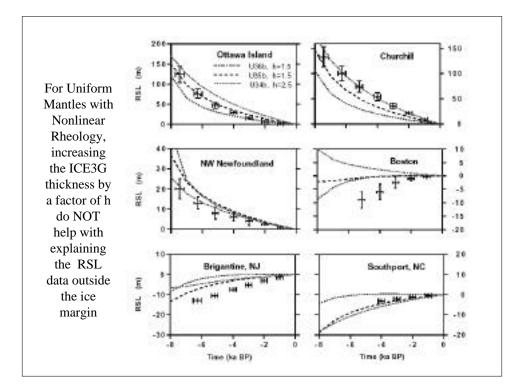
Karato (1998): Since the strain due to GIA is orders of magnitude smaller than tectonic strain, there is no interaction between rebound & tectonic stress



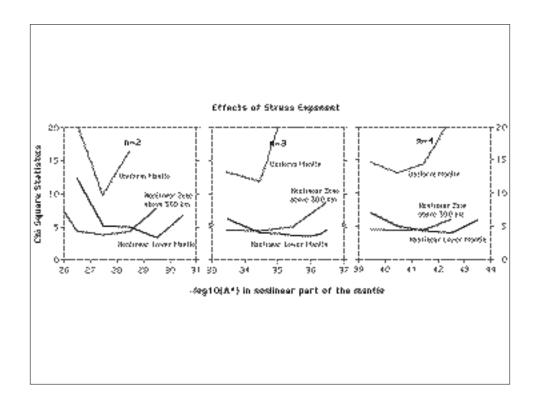


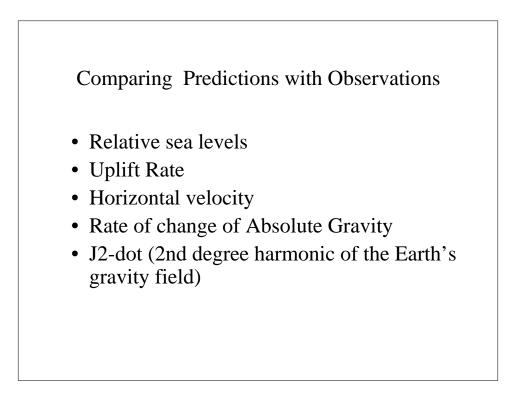


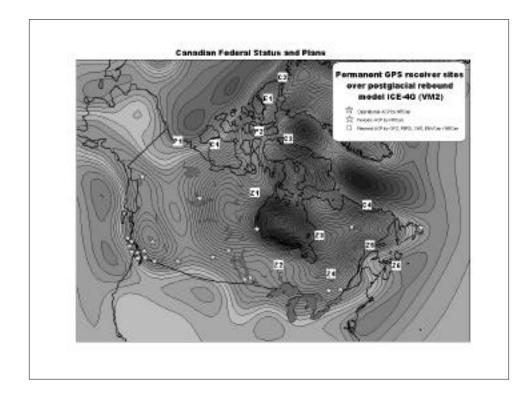


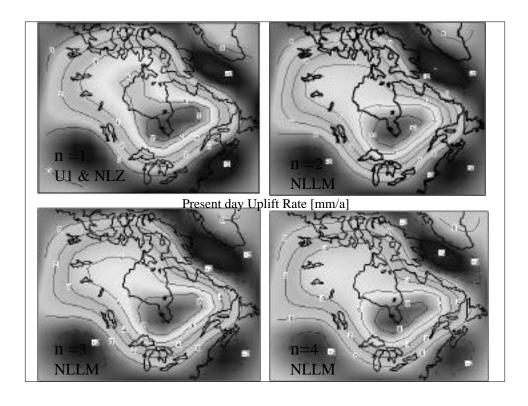


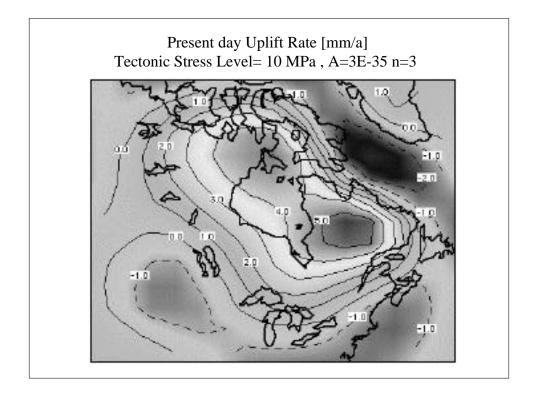
$$\chi^{2} = \frac{1}{M} \sum_{n=1}^{M} \left( \frac{\zeta_{observed} - \zeta_{predicted}}{s} \right)^{2}$$
$$\zeta = \text{sealevel height}$$
$$S = \text{standard deviation of error in height}$$
$$M = \text{number of sealevel observations}$$

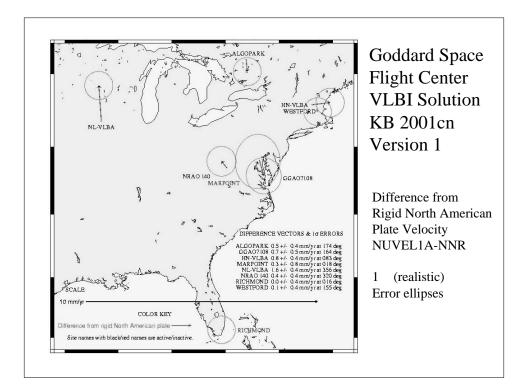


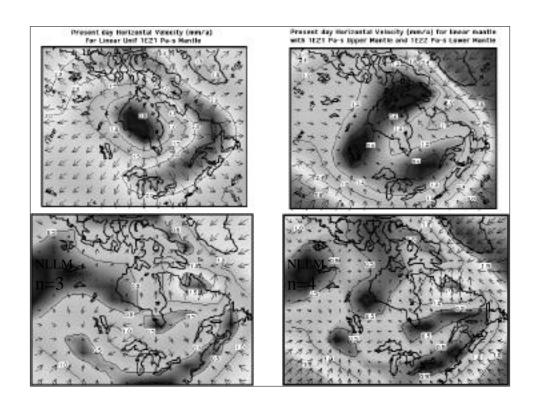


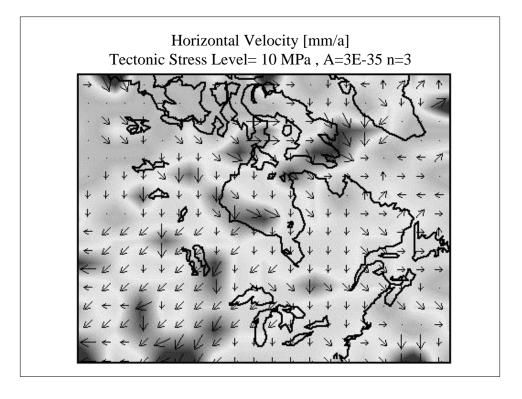


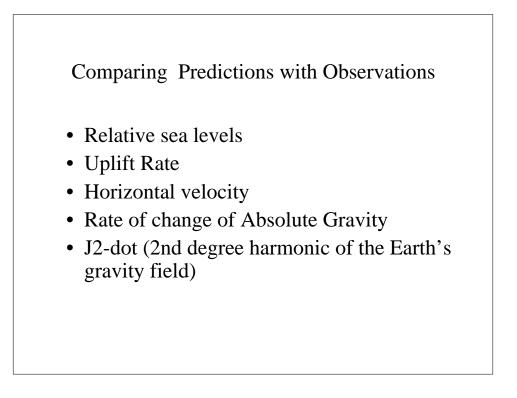


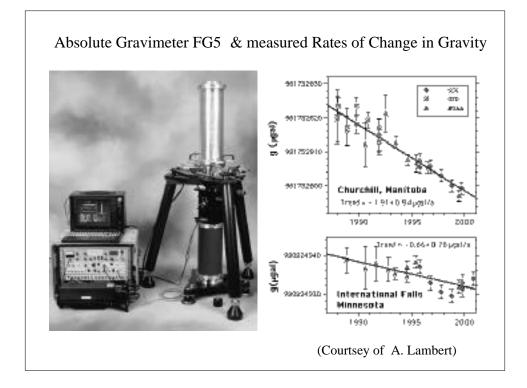


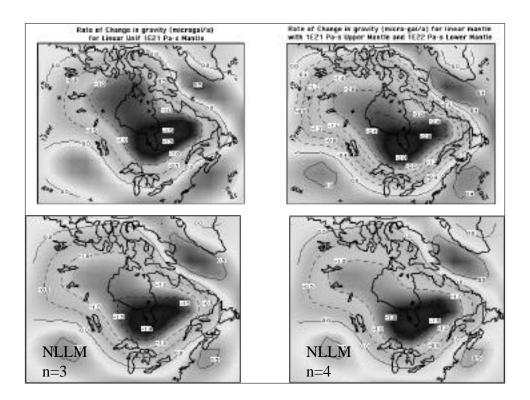


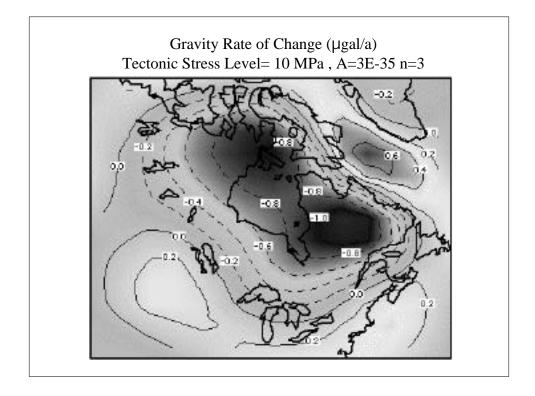


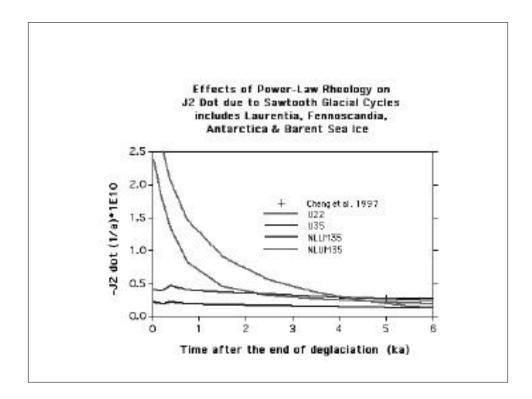


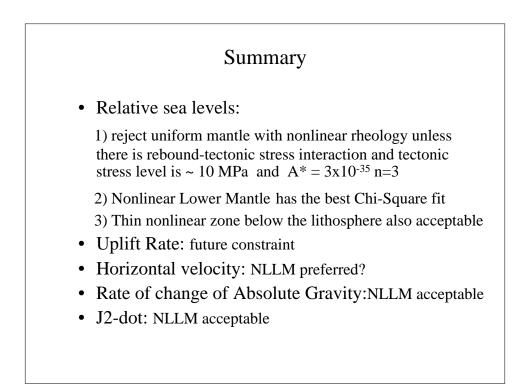












## Future Work

- For Surface Motion & Gravity, include:
  - a) Spherical Self-Gravitating Earth
  - b) Self-Gravitating sealevels
  - c) Compressibility
- For J2-dot, include :
  - a) Self-Gravitating sealevels
  - b) Compressibility
  - c) Recent melting events

